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Serial Number: 10/697,369

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Day : Wednesday

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**PALM INTRANET**

Inventor Information for 10/697369

Inventor Name	City	State/Country
LIANG, GEORGE	PALM CITY	FLORIDA

Appln Info	Contents	Petition Info	Atty/Agent Info	Continuity/Reexam	Foreign I
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1	US 7217097 B2	20070515	Cooling system with internal flow guide within a turbine blade of a turbine engine	416/97R		Liang; George
1	US 7195458 B2	20070327	Impingement cooling system for a turbine blade	416/97R		Liang; George
1	US 7189060 B2	20070313	Cooling system including mini channels within a turbine blade of a turbine engine	416/97R	415/115	Liang; George
1	US 7186089 B2	20070306	Cooling system for a platform of a turbine blade	416/193A		Liang; George
1	US 7163373 B2	20070116	Vortex dissipation device for a cooling system within a turbine blade of a turbine engine	416/97R	415/115	Liang; George
1	US 7137780 B2	20061121	Internal cooling system for a turbine blade	416/90R	416/97R	Liang; George
1	US 7137779 B2	20061121	Gas turbine airfoil leading edge cooling	416/1	416/97R	Liang; George
1	US 7128533 B2	20061031	Vortex cooling system for a turbine blade	416/97R		Liang; George
1	US 7118337 B2	20061010	Gas turbine airfoil trailing edge corner	416/1	416/97R	Liang; George
1	US 7118326 B2	20061010	Cooled gas turbine vane	415/115	416/97R	Liang; George
1	US 7114923 B2	20061003	Cooling system for a showerhead of a turbine blade	416/97R	415/115	Liang; George

1	US 7097417 B2	20060829	Cooling system for an airfoil vane	415/115		Liang; George
1	US 7090461 B2	20060815	Gas turbine vane with integral cooling flow control system	415/115		Liang; George
1	US 7029235 B2	20060418	Cooling system for a tip of a turbine blade	416/92	416/232; 416/97R	Liang; George
1	US 6994521 B2	20060207	Leading edge diffusion cooling of a turbine airfoil for a gas turbine engine	416/97R		Liang; George
1	US 6981846 B2	20060103	Vortex cooling of turbine blades	416/97R		Liang; George
1	US 6971851 B2	20051206	Multi-metered film cooled blade tip	416/1	416/97R	Liang; George
1	US 6955525 B2	20051018	Cooling system for an outer wall of a turbine blade	416/97R		Liang; George
1	US 6939102 B2	20050906	Flow guide component with enhanced cooling	415/115	415/116; 416/92; 416/97R	Liang; George
1	US 6932573 B2	20050823	Turbine blade having a vortex forming cooling system for a trailing edge	416/97R	415/115	Liang; George
1	US 6918742 B2	20050719	Combustion turbine with airfoil having multi-section diffusion cooling holes and methods of making same	415/115	416/97R	Liang; George
1	US 6916150	20050712	Cooling system for a tip of a	415/115	416/97R	Liang; George

	B2		turbine blade			
1	US 6902372 B2	20050607	Cooling system for a turbine blade	415/115	416/96R	Liang; George
1	US 6869268 B2	20050322	Combustion turbine with airfoil having enhanced leading edge diffusion holes and related methods	415/115	416/97R	Liang; George
1	US 6808367 B1	20041026	Cooling system for a turbine blade having a double outer wall	416/97R	415/115; 415/116	Liang; George
1	US 6616406 B2	20030909	Airfoil trailing edge cooling construction	416/97R		Liang; George
1	US 6602052 B2	20030805	Airfoil tip squealer cooling construction	416/97R	416/224	Liang; George
1	US 6471480 B1	20021029	Thin walled cooled hollow tip shroud	416/97R	415/115; 416/191; 416/233	Balkcum, III; J. Tyson et al.
1	US 6301877 B1	20011016	Ejector extension cooling for exhaust nozzle	60/226.1	239/127.3; 60/266; 60/39.5	Liang; George P. et al.
1	US 6210111 B1	20010403	Turbine blade with platform cooling	416/97R	415/115; 416/96R; 416/97A	Liang; George P.
1	US 6164059 A	20001226	Multi-expansion ejector nozzle with diverging walls	60/230	239/127.3; 239/265.17; 239/265.35; 60/266	Liang; George P. et al.
1	US 6139269 A	20001031	Turbine blade with multi-pass cooling and cooling air addition	416/97R	415/115	Liang; George P.
1	US 5975851 A	19991102	Turbine blade with trailing edge root	416/97R	415/115	Liang; George P.

			section cooling			
1	US 5665623 A	19970909	Method of fabricating totally self-aligned contacts for dynamic randomaccess memory cells	438/239	148/DIG.20; 257/E21.654; 438/297; 438/453; 438/666	Liang; George Wen Jya et al.
1	US 5605046 A	19970225	Cooled liner apparatus	60/752	60/266	Liang; George P.
1	US 5407133 A	19950418	Cooled thin metal liner	239/127.3	60/757	Liang; George P.
1	US 5157917 A	19921027	Gas turbine engine cooling air flow	60/226.1	60/226.3; 60/761	Liang; George P. et al.
1	US 5077969 A	19920107	Cooled liner for hot gas conduit	60/766	60/757; 60/759	Liang; George P. et al.
1	US 5056307 A	19911015	Exhaust nozzle cooling utilizing total engine flow	60/226.1	60/262; 60/39.5	Liang; George
1	US 5039562 A	19910813	Method and apparatus for cooling high temperature ceramic turbine blade portions	427/276	415/115; 427/336; 427/355; 427/427	Liang; George P.
1	US 5030060 A	19910709	Method and apparatus for cooling high temperature ceramic turbine blade portions	415/115	416/95	Liang; George P.
1	US 4752186 A	19880621	Coolable wall configuration	416/97R		Liang; George P.
1	US 4752184 A	19880621	Self-locking outer air seal with full backside cooling	415/116	415/173.3; 415/180	Liang; George
1	US 4601638	19860722	Airfoil trailing edge cooling	416/97R	415/115	Hill; Edward

	A		arrangement			C. et al.
1	US 4407632 A	19831004	Airfoil pedestaled trailing edge region cooling configuration	416/97R	415/115	Liang; George P.
1	US 3727070 A	19730410	REMOTE CONTROL AUTOMOBILE ENGINE STARTING APPARATUS	290/38R	290/37R	Liang; George H. K.